

A collection of historical artifacts is arranged on a light-colored surface. In the top left, a portion of a wooden chessboard with a checkered pattern and several chess pieces is visible. Below the chessboard, there are two medals: one with a red ribbon and a circular emblem, and another with a blue ribbon and a circular emblem. A pair of round-rimmed glasses with thin metal frames lies diagonally across the center. In the bottom left corner, a small, round, silver-colored compass with a white face and black markings is partially visible.

Localization with GPS

From GPS Theory and
Practice Fifth Edition
Presented by Martin
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Introduction

- ◆ GPS = Global Positioning System
- ◆ Three segments:
 1. Space (24 satellites)
 2. Control (DOD)
 3. User (civilian and military receivers)



GPS Overview

- ◆ Satellites transmit L1 and L2 signals
- ◆ L1--two pseudorandom noise signals
 - Protected (P-)code
 - Course acquisition (C/A) code (most civilian receivers)
- ◆ L2--P-code only
- ◆ Anti-spoofing adds noise to the P-code, resulting in Y-code



Observables

- ◆ Code pseudoranges

$$R = c \Delta t = c \Delta t(\text{GPS}) + c \Delta \delta = \varrho + c \Delta \delta.$$

$$\begin{aligned} \varrho &= \varrho(t^S, t_R) = \varrho(t^S, (t^S + \Delta t)) \\ &= \varrho(t^S) + \dot{\varrho}(t^S) \Delta t \end{aligned}$$



Observables

- ◆ Phase pseudoranges
 - N = number of cycles between satellite and receiver

$$\Phi = \frac{1}{\lambda} \rho + \frac{c}{\lambda} \Delta\delta + N$$



Observables

- ◆ Doppler Data
 - Dots indicate derivatives wrt time.

$$D = \lambda \dot{\Phi} = \dot{\varrho} + c \Delta \dot{\delta}$$



Observables

◆ Biases and Noise

Table 6.1. Range biases

Source	Effect
Satellite	Clock bias
	Orbital errors
Signal propagation	Ionospheric refraction
	Tropospheric refraction
Receiver	Antenna phase center variation
	Clock bias
	Multipath



Combining Observables

- ◆ Generally
- ◆ Linear combinations with integers
- ◆ Linear combinations with real numbers
- ◆ Smoothing



Mathematical Models for Positioning

- ◆ Point positioning
- ◆ Differential positioning
 - With code ranges
 - With phase ranges
- ◆ Relative positioning
 - Single differences
 - Double differences
 - Triple differences



Point Positioning

With Code Ranges

$$R_i^j(t) + c \delta^j(t) = \rho_i^j(t) + c \delta_i(t) .$$

With Carrier Phases

$$\Phi_i^j(t) + f^j \delta^j(t) = \frac{1}{\lambda} \rho_i^j(t) + N_i^j + f^j \delta_i(t) .$$

With Doppler Data

$$D_i^j(t) = \dot{\rho}_i^j(t) + c \Delta \dot{\delta}_i^j(t)$$



Differential Positioning

Two receivers used:

- Fixed, A: Determines PRC and RRC
- Rover, B: Performs point pos'ning with PRC and RRC from A

With Code Ranges

$$R_A^j(t_0) = \varrho_A^j(t_0) + \Delta\varrho_A^j(t_0) + \Delta\varrho^j(t_0) + \Delta\varrho_A(t_0)$$

$$R_B^j(t)_{\text{corr}} = \varrho_B^j(t) + \Delta\varrho_{AB}(t)$$



Differential Positioning

With Phase Ranges

$$\begin{aligned}\lambda \Phi_A^j(t_0) &= \varrho_A^j(t_0) + \Delta\varrho_A^j(t_0) + \Delta\varrho^j(t_0) + \Delta\varrho_A(t_0) + \lambda N_A^j \\ \lambda \Phi_B^j(t)_{\text{corr}} &= \varrho_B^j(t) + \Delta\varrho_{AB}(t) + \lambda N_{AB}^j\end{aligned}$$



Relative Positioning

Aim is to determine the baseline vector A->B.

A is known,

B is the reference point

Assumptions: A, B are simultaneously observed

Single Differences:

- two points and one satellite
- Phase equation of each point is differenced to yield

$$\Phi_{AB}^j(t) = \frac{1}{\lambda} \varphi_{AB}^j(t) + N_{AB}^j + f^j \delta_{AB}(t)$$



Relative Positioning

- ◆ Double differences
 - Two points and two satellites
 - Difference of two single-differences gives

$$\Phi_{AB}^{jk}(t) = \frac{1}{\lambda} \varphi_{AB}^{jk}(t) + N_{AB}^{jk}.$$



Relative Positioning

- ◆ Triple-Differences
 - Difference of double-differences across two epochs

$$\Phi_{AB}^{jk}(t_{12}) = \frac{1}{\lambda} \varrho_{AB}^{jk}(t_{12})$$



Adjustment of Mathematical Models

- ◆ Models above need adjusting so that they are in a linear form.
- ◆ Idea is to linearize the distance metrics which carry the form:

$$\begin{aligned} \varrho_i^j(t) &= \sqrt{(X^j(t) - X_i)^2 + (Y^j(t) - Y_i)^2 + (Z^j(t) - Z_i)^2} \\ &\equiv f(X_i, Y_i, Z_i) \end{aligned}$$



Adjustment of Mathematical Models

- ◆ Each coordinate is decomposed as follows:

$$X_i = X_{i0} + \Delta X_i$$

$$Y_i = Y_{i0} + \Delta Y_i$$

$$Z_i = Z_{i0} + \Delta Z_i$$

Allowing the Taylor series expansion of f

$$f(X_i, Y_i, Z_i) \equiv f(X_{i0} + \Delta X_i, Y_{i0} + \Delta Y_i, Z_{i0} + \Delta Z_i)$$

$$\begin{aligned} &= f(X_{i0}, Y_{i0}, Z_{i0}) + \frac{\partial f(X_{i0}, Y_{i0}, Z_{i0})}{\partial X_{i0}} \Delta X_i \\ &+ \frac{\partial f(X_{i0}, Y_{i0}, Z_{i0})}{\partial Y_{i0}} \Delta Y_i + \frac{\partial f(X_{i0}, Y_{i0}, Z_{i0})}{\partial Z_{i0}} \Delta Z_i + \dots \end{aligned}$$



Adjustment of Mathematical Models

- ◆ Computing the partial derivatives and substituting preliminary equations yields the linear equation:

$$\begin{aligned} \varrho_i^j(t) = & \varrho_{i0}^j(t) - \frac{X^j(t) - X_{i0}}{\varrho_{i0}^j(t)} \Delta X_i - \frac{Y^j(t) - Y_{i0}}{\varrho_{i0}^j(t)} \Delta Y_i \\ & - \frac{Z^j(t) - Z_{i0}}{\varrho_{i0}^j(t)} \Delta Z_i \end{aligned}$$



Linear Models

◆ Point Positioning with Code Ranges

- Recall: $R_i^j(t) = \rho_i^j(t) + c \delta_i(t) - c \delta^j(t)$
- Substitution of the linearized term (prev. slide) and rearranging all unknowns to the left, gives:

$$R_i^j(t) - \rho_{i0}^j(t) + c \delta^j(t) = -\frac{X^j(t) - X_{i0}}{\rho_{i0}^j(t)} \Delta X_i \\ - \frac{Y^j(t) - Y_{i0}}{\rho_{i0}^j(t)} \Delta Y_i - \frac{Z^j(t) - Z_{i0}}{\rho_{i0}^j(t)} \Delta Z_i + c \delta_i(t)$$



Linear Models

- ◆ Point Positioning with Code Ranges
- ◆ Four unknowns implies the need for four satellites. Let:

$$\ell^j = R_i^j(t) - \rho_{i0}^j(t) + c \delta^j(t)$$

$$a_{X_i}^j = - \frac{X^j(t) - X_{i0}}{\rho_{i0}^j(t)}$$

$$a_{Y_i}^j = - \frac{Y^j(t) - Y_{i0}}{\rho_{i0}^j(t)}$$

$$a_{Z_i}^j = - \frac{Z^j(t) - Z_{i0}}{\rho_{i0}^j(t)}$$



Linear Models

- ◆ Point Positioning with Code Ranges
- ◆ Assuming satellites numbered from 1 to 4

$$\ell^1 = a_{X_i}^1 \Delta X_i + a_{Y_i}^1 \Delta Y_i + a_{Z_i}^1 \Delta Z_i + c \delta_i(t)$$

$$\ell^2 = a_{X_i}^2 \Delta X_i + a_{Y_i}^2 \Delta Y_i + a_{Z_i}^2 \Delta Z_i + c \delta_i(t)$$

$$\ell^3 = a_{X_i}^3 \Delta X_i + a_{Y_i}^3 \Delta Y_i + a_{Z_i}^3 \Delta Z_i + c \delta_i(t)$$

$$\ell^4 = a_{X_i}^4 \Delta X_i + a_{Y_i}^4 \Delta Y_i + a_{Z_i}^4 \Delta Z_i + c \delta_i(t)$$

Superscripts denote satellite numbers, not indices.



Linear Models

Point Positioning with Code Ranges

- We can now express the model in matrix form as
 $l = Ax$ where

$$\underline{A} = \begin{bmatrix} a_{X_i}^1 & a_{Y_i}^1 & a_{Z_i}^1 & c \\ a_{X_i}^2 & a_{Y_i}^2 & a_{Z_i}^2 & c \\ a_{X_i}^3 & a_{Y_i}^3 & a_{Z_i}^3 & c \\ a_{X_i}^4 & a_{Y_i}^4 & a_{Z_i}^4 & c \end{bmatrix} \quad \underline{x} = \begin{bmatrix} \Delta X_i \\ \Delta Y_i \\ \Delta Z_i \\ \delta_i(t) \end{bmatrix} \quad \underline{\ell} = \begin{bmatrix} \ell^1 \\ \ell^2 \\ \ell^3 \\ \ell^4 \end{bmatrix}$$



Linear Models

Point Positioning with Carrier Phases

- Similarly computed.
- Ambiguities in the model raise the number of unknowns from 4 to 8
- Need three epochs to solve the system. It produces 12 equations with 10 unknowns.

Linear Models

Point Positioning with Carrier Phases

Linear Model

$$\lambda \Phi_i^j(t) - \varrho_{i0}^j(t) + c \delta^j(t) = - \frac{X^j(t) - X_{i0}}{\varrho_{i0}^j(t)} \Delta X_i - \frac{Y^j(t) - Y_{i0}}{\varrho_{i0}^j(t)} \Delta Y_i - \frac{Z^j(t) - Z_{i0}}{\varrho_{i0}^j(t)} \Delta Z_i + \lambda N_i^j + c \delta_i(t)$$

Linear Models

Point Positioning with Carrier Phases

$$l = Ax$$

$$\underline{l} = \begin{bmatrix} \lambda \Phi_i^1(t) - \varrho_{i0}^1(t) + c \delta^1(t) \\ \lambda \Phi_i^2(t) - \varrho_{i0}^2(t) + c \delta^2(t) \\ \lambda \Phi_i^3(t) - \varrho_{i0}^3(t) + c \delta^3(t) \\ \lambda \Phi_i^4(t) - \varrho_{i0}^4(t) + c \delta^4(t) \end{bmatrix}$$

$$\underline{A} = \begin{bmatrix} a_{X_i}^1(t) & a_{Y_i}^1(t) & a_{Z_i}^1(t) & \lambda & 0 & 0 & 0 & c \\ a_{X_i}^2(t) & a_{Y_i}^2(t) & a_{Z_i}^2(t) & 0 & \lambda & 0 & 0 & c \\ a_{X_i}^3(t) & a_{Y_i}^3(t) & a_{Z_i}^3(t) & 0 & 0 & \lambda & 0 & c \\ a_{X_i}^4(t) & a_{Y_i}^4(t) & a_{Z_i}^4(t) & 0 & 0 & 0 & \lambda & c \end{bmatrix}$$

$$\underline{x}^T = \left[\Delta X_i \quad \Delta Y_i \quad \Delta Z_i \quad N_i^1 \quad N_i^2 \quad N_i^3 \quad N_i^4 \quad \delta_i(t) \right]$$



Linear Models

Relative Positioning

- Carrier phases considered
- Double-differences treated
- Recall: DD equation * _

$$\lambda \Phi_{AB}^{jk}(t) = \varphi_{AB}^{jk}(t) + \lambda N_{AB}^{jk}$$

- The second term on the lhs is expanded and linearized as in previous models to yield:

Linear Models

Relative Positioning

- The second term on the lhs is expanded and linearized as in previous models to yield ([9.133]...see paper pg 262)
- l 's:

$$\begin{aligned} \ell_{AB}^{jk}(t) = & a_{X_A}^{jk}(t) \Delta X_A + a_{Y_A}^{jk}(t) \Delta Y_A + a_{Z_A}^{jk}(t) \Delta Z_A \\ & + a_{X_B}^{jk}(t) \Delta X_B + a_{Y_B}^{jk}(t) \Delta Y_B + a_{Z_B}^{jk}(t) \Delta Z_B + \lambda N_{AB}^{jk} \end{aligned}$$

Linear Models

Relative Positioning

- The right hand side is abbreviated as follows (a's):

$$a_{Y_A}^{jk}(t) = + \frac{Y^k(t) - Y_{A0}}{\varrho_{A0}^k(t)} - \frac{Y^j(t) - Y_{A0}}{\varrho_{A0}^j(t)}$$

$$a_{Z_A}^{jk}(t) = + \frac{Z^k(t) - Z_{A0}}{\varrho_{A0}^k(t)} - \frac{Z^j(t) - Z_{A0}}{\varrho_{A0}^j(t)}$$

$$a_{X_B}^{jk}(t) = - \frac{X^k(t) - X_{B0}}{\varrho_{B0}^k(t)} + \frac{X^j(t) - X_{B0}}{\varrho_{B0}^j(t)}$$

$$a_{Y_B}^{jk}(t) = - \frac{Y^k(t) - Y_{B0}}{\varrho_{B0}^k(t)} + \frac{Y^j(t) - Y_{B0}}{\varrho_{B0}^j(t)}$$

$$a_{Z_B}^{jk}(t) = - \frac{Z^k(t) - Z_{B0}}{\varrho_{B0}^k(t)} + \frac{Z^j(t) - Z_{B0}}{\varrho_{B0}^j(t)}$$



Linear Models

Relative Positioning

- Since the coordinates of A must be known, the number of unknowns is reduced by three. Now, 4 satellites (j, k, l, m) and two epochs are needed to solve the system.



Extra References

- ◆ Introduction and overview:
<http://www.gpsy.org/gpsinfo/gps-faq.txt>